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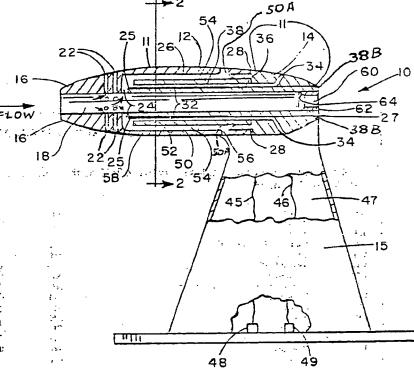
#### **Published**

With international search report

(54) Title: FLOW DEVICE FOR SENSORS

### (57) Abstract

In the measurement of parameters such as temperature and humidity or the like of a high velocity fluid, errors attributed to the velocity are encountered. The device of the invention protects sensors from direct impingment by such high velocity fluid by providing a housing (14) having a leading portion (16) and a trailing por-FLOW tion (14) forming the elongated longitudinal housing (11) with a bore (18) extending longitudinally through the portions for flow therethrough. The flow rate is internally regulated to a desired rate by a restriction (60) in the bore located in the trailing portion. Where the fluid stream is air, the restriction provides for internal Mach number regulation at various flow rates. A first plurality of ports (22) is located in the leading portion for boundary layer control. The device has one or more annular fluid passageways (24) downstream of the first ports coupled to the bore and ported to the exterior to cooperated with the bore to provide a lower pressure area for flow. therethrough, and a sensor (38) disposed in at least one such passageway. The leading portion may incorporate anti-icing for air stream measurement. For temperature sensors, radiation: shielding (50) is provided.



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# FLOW DEVICE FOR SENSORS BACKGROUND OF THE INVENTION

1. Field of the Invention

The invention relates to measurement of flow parameters and provides a desired flow to impinge on the sensing element at various stream flow rates.

2. Description of the Prior Art

Devices for sensing data in a flow stream which provide for accurate, fast response and ruggedness at relatively low cost are desirable. While certain devices may have been useful in the past, the advantages of the present device, as disclosed herein, are apparent.

In the prior art, United States Patent 15 2,970,475 issued February 7, 1961 and has the same assignee as the present invention. That patent provided for boundary layer control (BLC) for a gas temperature probe, thus increasing the capability of a temperature sensor to accurately read the total temperature of the 20 fluid when brought to rest against the sensing element. In United States Patent 3,512,414, also held by the same assignee as the present invention, further advantages of ruggedness, minimum drag, BLC and others are shown. utility of inventions for such applications has been 25 established by use for laboratory testing, and by extensive commercialization for air vehicles.

Recently, United States Patent 4,152,938 issued. While this patent may appear to be somewhat similar to the present invention, it is stated at column 5, line 61, "Because the probe requires unobstructed flow through bore 14 the preferred structure...". As herein disclosed, regulation of internal flow enhances operation for example, where the fluid is air, regulation of flow enhances operation at Mach numbers above 0.5. No means, method or the desirability of such means or method for regulation is disclosed in United States Patent 4,152,938. Other



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advantages are apparent from the invention as herein disclosed.

### SUMMARY OF THE INVENTION

A device for measuring a parameter in a fluid 5 stream, which, in one preferred embodiment, is temperature in an air stream, is disclosed. The device has a leading portion and a trailing portion thus forming an elongated longitudinal housing with a bore substantially centered on its longitudinal axis for flow 10 of the fluid stream therethrough. A restriction or throat is disposed in the bore at the trailing portion to control the rate of flow through the device. the fluid stream is air, Mach number regulation in the bore is thus provided. Regulation of the rate of flow 15 internal to the device enhances operation at higher ambient flow rates. For example, for operation in air at higher subsonic Mach numbers, e.g. above Mach .5, poor overall performance such as poor accuracy and repeatability, is avoided. Selection of a restriction or throat 20 size based on the bore diameter to regulate the internal rate of flow as for air to a Mach number below .5 has resulted in enhanced performance. Further, the restriction or throat raises the internal pressure so that routing of a portion of undisturbed fluid to the sensor, 25 which is disposed in an annular passageway outboard of the bore, is more easily accomplished. Because of porting of fluid from the bore for boundary layer control, the sensing element, disposed in the passageway, is subjected to relatively undisturbed, free stream fluid 30 parameters such as temperature which, for example, is raised substantially adiabatically upon impingement on the sensing element. Radiation shielding for the sensor is provided. A further advantage is that since the internal pressure is raised due to the restriction or 35 throat, it is not required that the external shape of the housing increase in diameter rearwardly as required



in United States Patent 4,152,938. The internal pressure is raised considerably above the local static pressure and, hence, the flow is more easily induced into the passageway and to the sensor. Similarly, the raised internal pressure compared to the local static pressure enables porting of the first passageway to the exterior of the device from the area of constant diameter, at the device strut or in an aft facing direction as desired.

It is understood herein that the fluid stream may be moving with relation to the device, which is in a fixed location such as a gas conduit, wind tunnel or the like, or the device may be mounted to an air vehicle, such as aircraft, missile, helicopter, or the like and such air vehicle may be moving through the fluid or air mass. The terms "Mach" and "total temperature", "flow", "pressure" and others used herein have the same meanings as in the aeronautical sciences.

## BRIEF DESCRIPTION OF THE DRAWINGS

Figure 1 is a sectional view of a first preferred embodiment of the housing taken as on line 1--1 in Figure 2 with a fragmentary view of the strut for mounting the device;

Figure 2 is a sectional view of the housing 25 taken as on line 2--2 in Figure 1;

Figure 3 is a fragmentary sectional view of a temperature sensing element for the device of the present invention;

Figure 4 is a fragmentary sectional view of
another preferred embodiment of the housing having
electrical anti-icing taken on the same line as Figure 1
and a fragmentary view of the housing; and

Figure 5 is a sectional view of a second preferred embodiment of the housing taken as on the same line as Figure 1 and a fragmentary view of the strut.



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### DESCRIPTION OF THE PREFERRED EMBODIMENT

A first preferred embodiment of a device for measuring a parameter in a fluid stream is shown in Figure 1 and Figure 2 and is generally indicated at 10. Device 10 includes a housing 11 which is formed of two main housing elements connected together, a leading portion 12 and a trailing portion 14 which form the elongated longitudinal shape of device 10. A strut 15 connected to trailing portion 14 supports device 10 in a desired position relative to a flow stream.

Leading portion 12 preferably is cast or formed from a cylindrical block of metal. A flow face 16 preferably is contoured to provide a relatively sleek aerodynamic shape and to reduce drag. A central, relatively uniform bore 18 is made substantially along the axis of the elongated longitudinal housing which housing is formed by leading portion 12 and trailing portion 14 for flow therethrough. Leading portion 12 has an inlet port for flow, which flow is indicated as left to right by the arrow in Figure 1. Trailing portion 14 through bore 18 is the main outlet for flow.

At the point on flow face 16 where the leading portion approaches its maximum exterior diameter a first series of ports 22, which preferably are two rows of six ports each spaced around leading portion 12, are provided and the ports extend through the wall of leading portion 12 to bore 18. These ports 22 cooperate with bore 18 to exhaust flow from bore 18 to the exterior, thus providing boundary layer control (BLC) in bore 18. This concept is fully explained in U.S. Patent 2,970,475.

Trailing portion 14 is also cast or formed from a metal block. Trailing portion 14 has a first end 25 and an outlet end 27. The first end 25 of trailing portion 14 is suitably dimensioned to be inserted into a large bore 26 in leading portion 12 thus forming an annular passageway open to bore 18 through an annular inlet 24. Leading portion 12 and



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trailing portion 14 preferably are connected together in a conventional manner as by welding, brazing or the like where surfaces of the two overlap as at 28. Ports 22 are spaced closely upstream from annular inlet 24 and the flow through ports 22 urges the lighter particles of the flow, for example, the molecules forming air, to change direction more than the heavier particles such as the suspended solids, raindroplets or the like. Similarly, the lighter particles tend to flow into the annular inlet 24 from bore 18.

A first annular passageway 32 is formed by casting or machining in trailing portion 14 radially outboard of bore 18. Passageway 32 has a plurality of ports 34, preferably six, which preferably are angled from the longitudinal or the flow axis. Ports 34 are located near the maximum exterior diameter of trailing portion 14, but may exhaust parallel to the longitudinal or flow axis. Passageway 32 then cooperates with bore 18, inlet 24 and ports 34 as well as ports 22 to provide a relatively undisturbed flow of fluid through passageway 32.

An inner surface 36 of passageway 32 preferably has a sensor or element 38 disposed thereon. example of a sensor is shown in Figure 3. An insulator layer 40 such as mica is first disposed on surface 36. An element 38, which preferably is a platinum wire resistor is then wrapped over insulator 40 and a second layer of an insulator 42 is then disposed on element 38. For further protection for element 38 an outer tube 44 preferably is slid or drawn over the entire element 38. Tube 44 may then be mechanically or hydrostatically swaged onto element 38 to further improve ruggedness and response time. Conventional treatment to hermetically seal element 38 preferably is also provided. Suitable leadwires 45 and 46 are then conventionally connected to ends of sensor 38 respectively in a flow direction



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longitudinally aft of sensor 38 in trailing portion 14. As best shown in Figure 1, leadwires 45 and 46 are then routed through a central cavity 47 in strut 15 to a desired interface connection 48 and 49 respectively. Other sensors such as pyrometers, thermocouples, humidity sensors, pressure sensors or the like and known mounting methods for such sensors may be used. Sensor 38 is mounted in position before housing portion 12 and 14 are joined together.

The ruggedness of a resistance wire wound over a sleeve is time tested and proven in years of applications in other temperature sensing applications. Sensing element 38 embodies this construction, which provides a substantially greater expected life than bare wire sensors. Response time and ruggedness of sensor 38 is improved by swaging tube 44.

An annular (tubular) separator 50 is preferably formed in trailing portion 14. Separator 50 provides an outboard barrier 52 for first passageway 32 and radiation shielding for sensor 38. Separator 50 in cooperation with the large bore 26 of leading portion 12 also forms a second annular passageway 54 in device 10. Second passageway 54 exhausts to the exterior through a plurality of ports 56, preferably six. Ports 56 preferably are located somewhat aft of the longitudinal midpoint of device 10. Passageway 54 in cooperation with the fluid stream in bore 18, inlet 24, and ports 56 provides a third path for the fluid stream flow. Passageway 54 further urges the flow into inlet 24 and provides further BLC so that the flow reaching sensor 38 is relatively undisturbed having substantially free stream flow characteristics. Exterior wall 58 surrounding large bore 26 provides a second radiation shield for sensing element To permit ease for mounting sensor 38 on inner surface 36, trailing portion 14 preferably is an assembly comprised of separator 50 which has a cylindrical shape



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and is brazed to trailing portion 14 at 50A, then the tube 38A which has sensor 38 mounted on its inner surface 36 preferably is brazed to trailing portion 14 at 38B.

A restriction or sonic throat 60 is formed at outlet end 27 of trailing portion 14 by welding or brazing a suitable metal insert 62 to the inner wall of bore 18, thus reducing the effective inside diameter of bore 18 at outlet end 27. Insert 62 on its flow facing side preferably is formed to have a taper 64 from bore 18 to the reduced diameter of restriction 60 to reduce turbulence in the flow caused by restriction 60. preferred embodiment for air flow, the diameter of bore 18 was 6 millimeters and the minimum throat 60 diameter was 5 millimeters and the maximum diameter of the device 18 millimeters, thus regulating the rate of flow of air for higher subsonic Mach numbers of air flow to Mach .35 in bore 18 of device 10. It is understood that other bore and throat diameters for regulation to other selected flow rates for other fluids or other Mach numbers for air are determined using conventional calculation methods.

In operation, restriction or throat 60 regulates rate of flow through device 10 and raises the internal pressure in bore 18 to further urge flow through both ports 22 and inlet 24, and through passageways 32 and 54. This increased internal pressure urging the flow through such ports and passageway enhances conditions so that fluid reaching sensor 38 has parameters substantially similar to the fluid stream in an undisturbed and uncontaminated state.

The construction of this embodiment both takes advantage of the prior art teachings by routing heavier particles in the fluid through bore 18, but also diverts only relatively undisturbed free stream fluid to the sensor 38. Restriction or throat 60 enhances the capability of device 10 to divert the fluid with minimal disturbance



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or turbulence added thereto.

In Figure 4 another preferred embodiment is There, flow face 16 of leading portion 12 is shown with electrical heating element 80 for anti-icing. Cavity 47 in strut 15 is shown with leadwires 45 and 46 and a heating element 80 therein. In one embodiment heating element 80 is comprised of two conductors with a resistance therebetween. The conductors are at different potentials and, hence a current flows through the resistor thus generating heat for anti-icing. Other conventional heating elements such as coils, resistors and the like may be used. As shown, heating element 80 preferably is conventionally disposed in flow face 16 in leading portion 12. Heating element 80 also is shown in cavity 47 near the leading edge of strut 15 to provide anti-icing on such leading edge, and may also be disposed in outlet end 27 or other areas of device 10 for anti-icing as desired.

In Figure 5 another preferred embodiment is 20 There, flow face 16 of leading portion 12 is shown as somewhat more blunt than as shown in Figure 1. In this embodiment, cavity 47 in strut 15 is shown with leadwires 45 and 46 and a conduit 90 for carrying hot gas, such as, engine bleed air from a jet engine compressor 25 of an air vehicle, or rocket exhaust gas or the like, which conduit is provided to conduct heat to flow face 16 for anti-icing. Conduit 90 preferably is routed on the leading edge of strut 15 to provide anti-icing on the surface also. Conduit 90 is coupled to an annular cavity 30 92 in leading portion 12 which provides the hot gas to heat flow face 16. The hot gas is then exhausted through a suitable configuration of hot gas ports 94. If desired, such hot gas may also be routed to outlet end 27 or other areas of housing 11.



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### WHAT I CLAIM IS:

 A device for measuring a parameter in a fluid stream comprising a housing having an exterior and having

flow regulation,

an elongated longitudinal shape and a bore along the shape for flow of the fluid stream therethrough, a leading portion at a first end of the bore, a trailing portion at a second end of the bore, restriction means in the bore for internal

a first plurality of port means from the bore to the exterior of the housing in the leading portion for providing boundary layer control,

first annular fluid passageway means in the housing downstream of the first port means coupled to and outboard of the bore and ported to the exterior to cooperate with the bore to provide a lower pressure area for the fluid which flows through the first passageway means, and sensor means to measure the parameter

sensor means to measure the parameter supported in the first passageway means.

- 2. Apparatus according to Claim 1 further comprising a second annular passageway means formed in the housing outwardly of the first passageway means coupled to the bore and to the first passageway means and ported to the exterior to enhance flow to the sensor, which flow has parameters substantially similar to the fluid stream in an undisturbed state.
- 3. Apparatus according to claim 2 wherein the restriction is of size to regulate internal flow rate to less than Mach .5.



- 4. Apparatus according to claim 2 wherein the restriction is of size to regulate internal flow rate to Mach .35.
- 5. Apparatus according to claim 2 wherein the leading portion has a blunt nose.
- 6. Apparatus according to claim 2 wherein the leading portion further comprises heating means for anti-icing.
- 7. Apparatus according to claim 2 wherein the wall forming the second passageway means is selected from a material to provide radiation shielding for the sensor means.
  - 8. Apparatus according to claim 2 wherein the sensor means is a platinum resistor.
- 9. Apparatus according to claim 2 wherein the sensor means has an internal element and an outer covering hermetically sealing the element.
  - 10. An air data sensing device comprising inlet means to provide for air flow to enter the device,

outlet means coupled to the inlet means to provide for air flow to exhaust from the device,

central passageway means in the inlet means and the outlet means for air flow therethrough,

the outlet means further having means defining a flow restriction in the central passageway means to regulate the Mach number of the flow therethrough,

first port means in the device open to the central passageway means and oriented to radially exhaust flow from the central passageway for Boundary Layer Control, first annular fluid flow passageway means in the device connected to the central passageway

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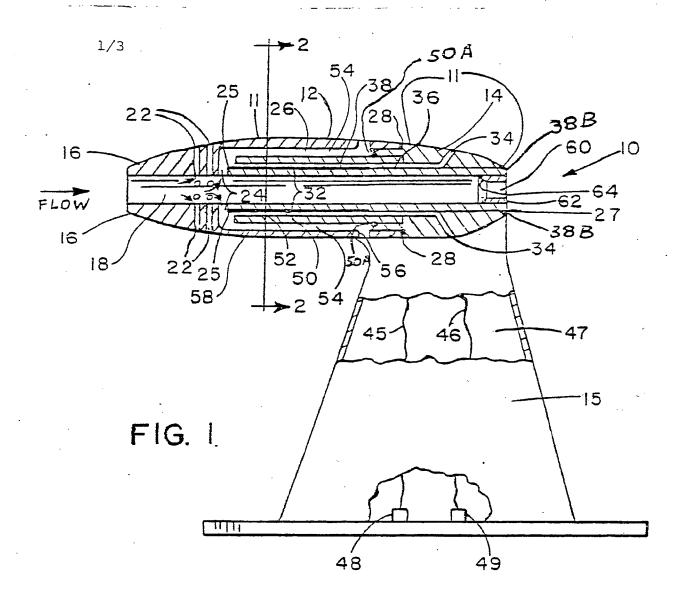
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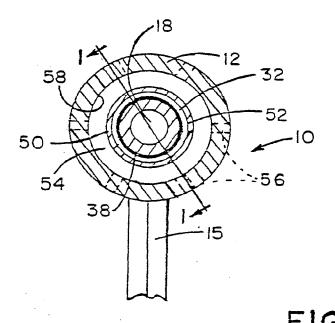


means downstream of the first port means, and exhausting fluid from the device for providing air flow having substantially free stream air characteristics, second annular fluid passageway means in said device outwardly of and fluidly coupled to the first annular passageway means and to the central passageway means and exhausting fluid from the device to further enhance Boundary Layer Control, sensor means annularly disposed in the first annular passageway means to measure the parameter.

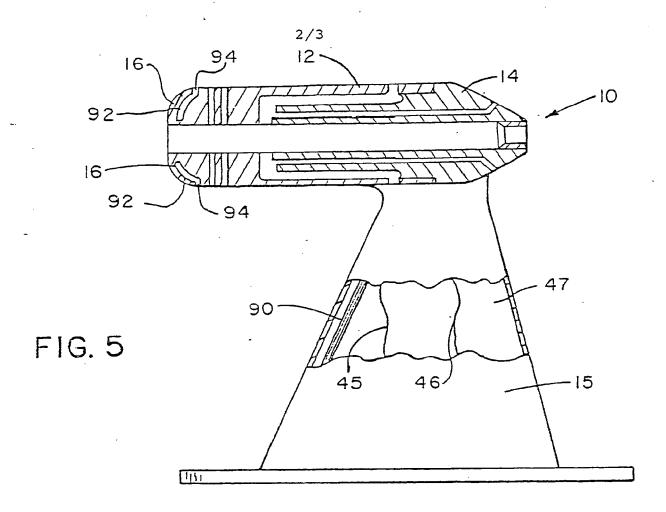
- 11. Apparatus according to claim 10 wherein the inlet means further comprises a relatively blunt nose on the flow face of this inlet means.
- 12. Apparatus according to claim 10 or 11 wherein the inlet means further comprises heater means for anti-icing.
- 13. Apparatus according to claim 10 or claim 11 wherein the restriction means regulates the Mach number.
- 14. Apparatus according to claim 10 or claim 11 wherein the restriction means regulates the Mach number to between .1 and .5 Mach.
- 15. Apparatus according to claim 10 or claim 11 wherein the restriction means regulates the Mach number to .35 Mach.
- 16. Apparatus according to claim 10 wherein the sensor means is a platinum resistor.
- 17. Apparatus according to claim 16 wherein the sensor means is swaged.

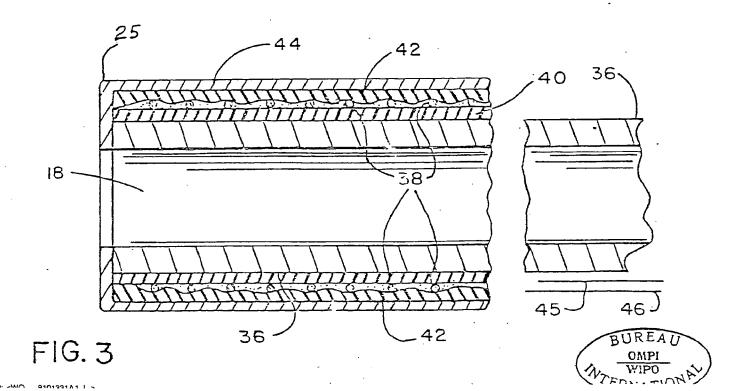


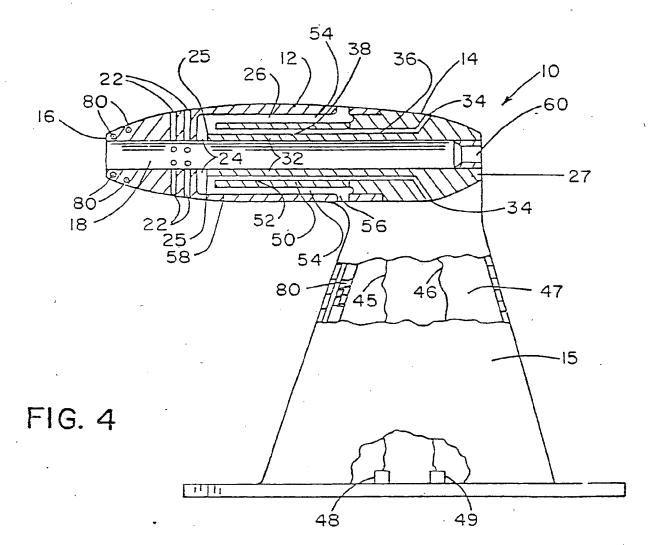














# INTERNATIONAL SEARCH REPORT

International Application No PCT/US80/01463

I. CLASSIFICATION OF SUBJECT MATTER (if several classification symbols apply, indicate all) 3					
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II. FIELDS SEARCHED					
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A	us,	А	2,970,475 Publish 1961 Werner	ned 7 February	
А	us,	Α.	3,512,414 Publish 1970 Rees	ned 19 May	
А	US,	А	4,152,938 Publis 1979 Danninger	hed 8 May	1-17
А	us,	Α	3,170,328 Publis 1965 Werner	hed 23 February	1-17
· A	US,	А	2,931,227 Publis 1960 Werner et al	hed 5 April	8,9,16,17
Α .	US,	А	2,588,840 Publis 1952 Howland	hed 11 March	5,6,11-15
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